

# Benchmarking Hytrac simulations of the Richtmyer-Meshkov Instability at First Light Fusion



first light

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## Motivation

- First Light Fusion (FLF) is investigating projectile-driven inertial confinement fusion.
- Projectiles are driven to velocities of 10-20 km/s by electromagnetic launch on our pulsed power machine M3.
- Hyper-velocity projectiles drive strong shocks across material interfaces in our targets.
- Therefore, the Richtmyer-Meshkov instability (RMI) may be an important factor in determining target performance.
- Modelling is performed using two in-house codes: Hytrac (2D AMR w/ front-tracking) and B (3D multimaterial resistive-MHD)
- RMI benchmarking is important for:
  - validation of hydrodynamic models
  - assessing RMI impact on target designs
- Our aim is to develop an RMI modelling capability valid over a range of target setups and suitable for reproducible CI testing.

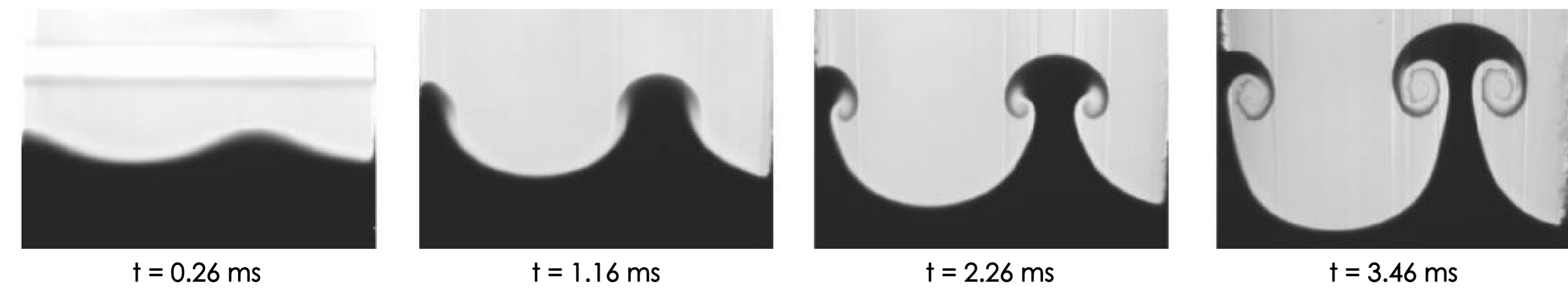


Fig. 1: (above) Experimental data (PLIF images) from the shocktube experiments of Jacobs & Krivets [3] illustrating the evolution of a single-mode Richtmyer-Meshkov instability; the images above show an Air-SF<sub>6</sub> interface for positive Atwood no. at  $M_i = 1.3$  and  $\lambda = 5.9$ cm. The instability typically evolves in four phases: (1) linear growth, (2) bubble/spike formation, (3) K-H instability seeding, and (4) fine-scale instability formation.

- The R-M instability [1] arises when fluid interfaces are impulsively accelerated, i.e., when shocks traverse material interfaces.
- We are benchmarking against experimental Air-SF<sub>6</sub> shocktube [2,3] and Be-AGAR NOVA laser results [4], due to the well-documented conditions and high quality data available.
- The Air-SF<sub>6</sub> test has  $A = 0.62$  and is shocked at  $M_i = 1.3$ . The Be-AGAR test has  $A = -0.87$  and  $M_i \approx 9-18$ . These conditions explore a diverse range of shock phenomena over different time/space scales.

$$A = \frac{\rho_{\text{ahead}} - \rho_{\text{behind}}}{\rho_{\text{ahead}} + \rho_{\text{behind}}}$$

- Instability character is defined by the Atwood number; Unlike the Rayleigh-Taylor instability (RTI), the RMI grows for both +ve and -ve A.

## Implementation of analytical RMI models

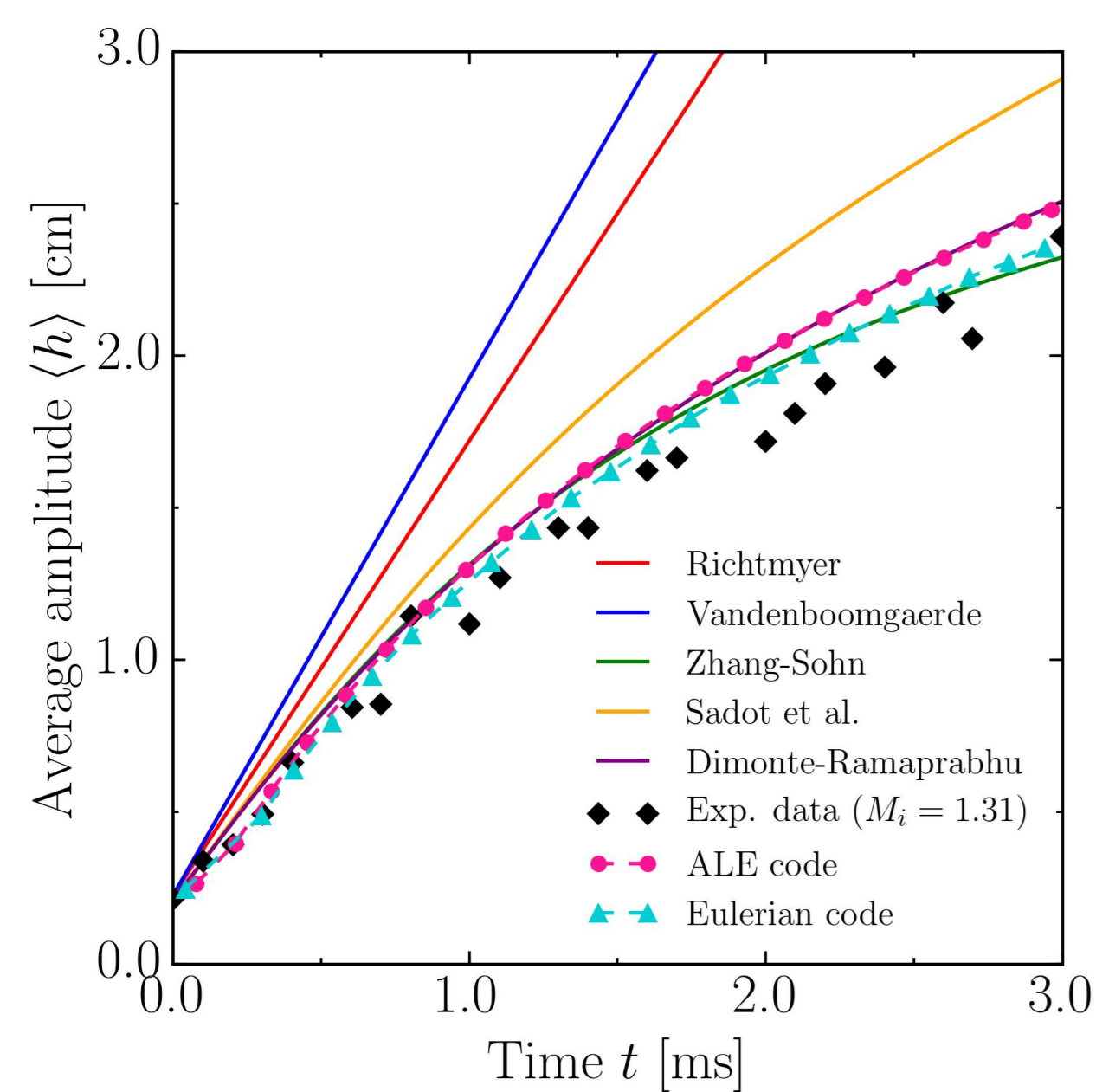


Fig. 2: Model comparison vs the Tri-Lab test suite RMI test, i.e., positive Atwood no. test. Data points digitised from [5]. The D-R and Zhang-Sohn models provide the best match to simulations and experimental data.

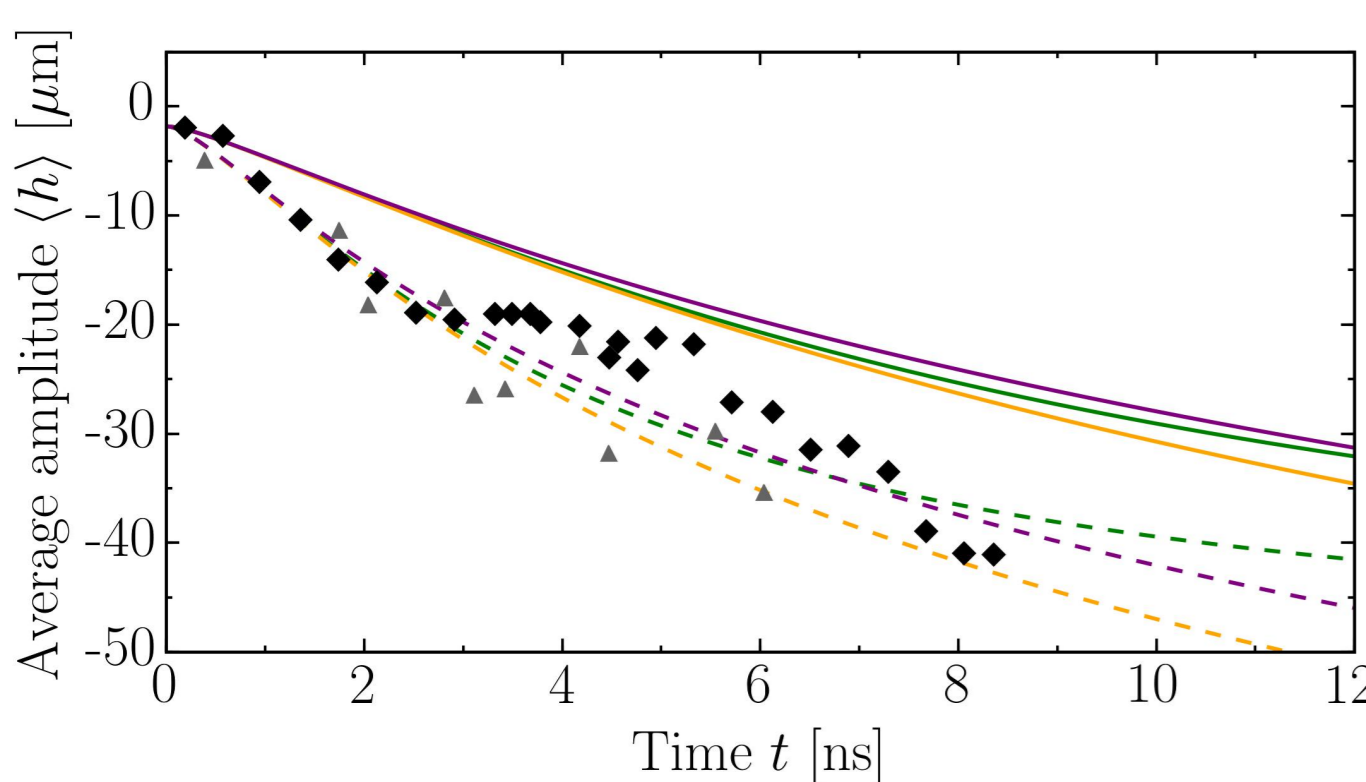


Fig. 4: A comparison of the non-linear models against the D-R NOVA data, as initialised w/ the Richtmyer (solid) and the Meyer-Blewett (dashed) models. The Meyer-Blewett-initialised models provide a better match to the data.

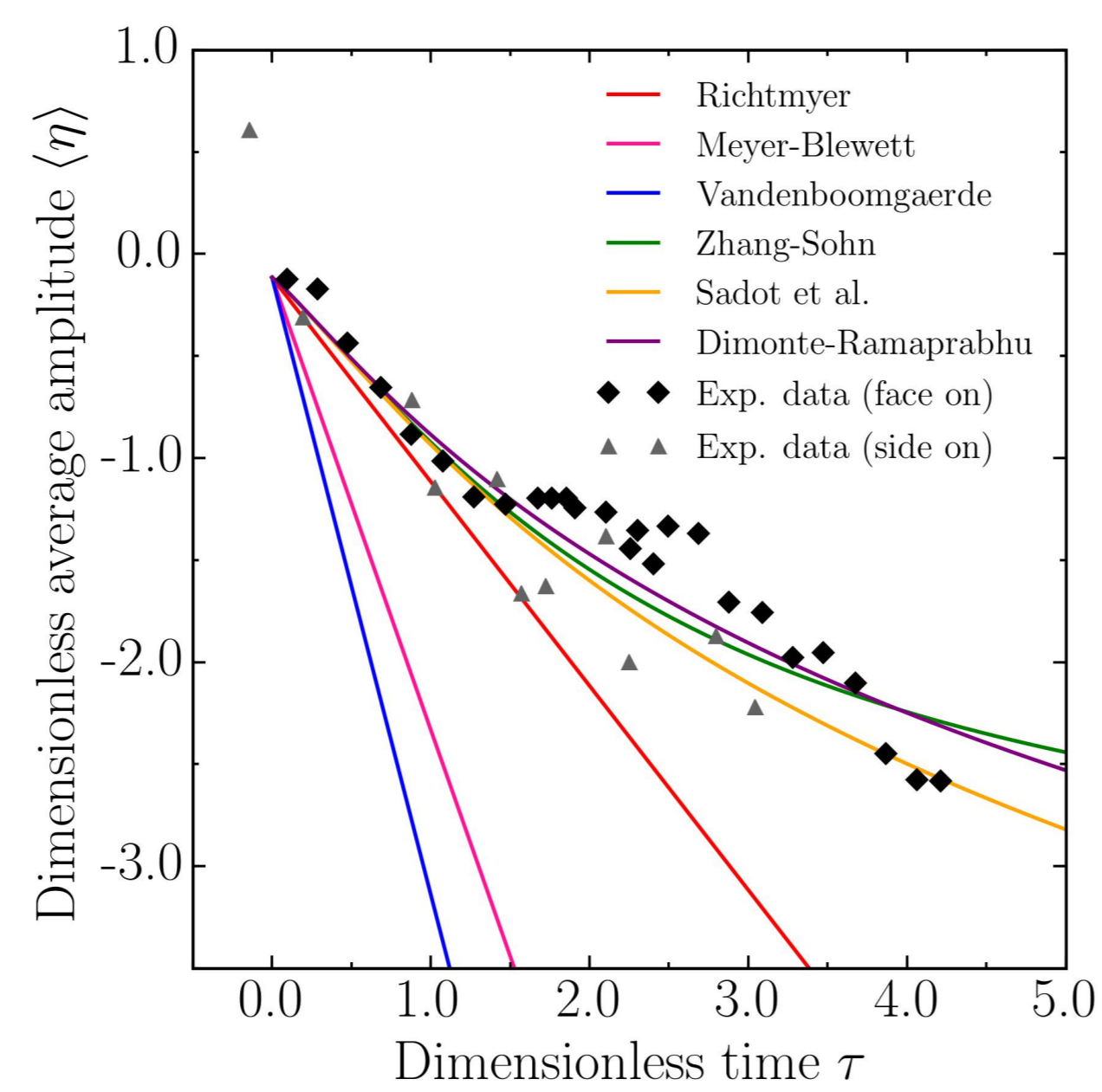


Fig. 3: Model comparison vs the Dimonte-Remington (fast) case, i.e., negative Atwood no. In this case the three non-linear models are all comparable to the experiment result, but the data spread is too large to clearly identify the best.

- Analytical models comprise three categories; impulsive, linear and non-linear (see Table 1).
- The simplest analytical model is the impulsive Richtmyer [1] setup - found from the impulsive limit of Taylor's linear RTI growth equation:
- The Tri-Lab test suite (TLTS) [5] RMI verification test has been performed (Fig. 2). This consists of a perturbed Air-SF<sub>6</sub> interface under conditions of [2], but with  $P = 20$  MPa.
- Fig. 3 compares models under experimental Dimonte-Remington [4] NOVA conditions.
- No significant difference was observed using Be Ideal gas EoS and FEOS.
- Fig. 4 shows the importance of the linear growth rate used to initialise the non-linear model.

Model	Validity / Notes
Richtmyer (R) [1]	Impulsive For reflected shocks (RS) - use post-shock values: $V_R(t) = ku_c h_0^+ A^+$ .
Meyer-Blewett (MB) [6]	Impulsive Valid for RS and reflected rarefactions: $V_{MB}(t) = \frac{1}{2}ku_c (h_0^+ + h_0^-) A^+$ .
Dimonte-Ramaprabhu (DR) [11]	Non-linear Non-linear model reproduces asymptotic growth rate at early and late times.

Table 1: A summary of different RMI models considered in this benchmarking work and their notable features.

- The Dimonte-Ramaprabhu growth rate is:

$$V_{b/s} = V_0 \frac{1 + [2 - F_{b/s}] \tau}{1 + C_{b/s} \tau + [2 - F_{b/s}] F_{b/s} \tau^2} \begin{cases} F_{b/s} (|A^+|) = 1 \pm |A^+| \\ C_{b/s} (|A^+|, |\eta_0^+|) = \frac{1}{4} \left[ \frac{7}{2} + F_{b/s} (|A^+|) - (3 - F_{b/s} (|A^+|)) |\eta_0^+| \right] \end{cases}$$

- Integrating yields the D-R model amplitude change:

$$\Delta \eta_{b/s}(t) = \text{sgn}(V_0) [a_{b/s} J(\tau; b_{b/s}, c_{b/s}) + b_{b/s} K(\tau; b_{b/s}, c_{b/s})] \begin{cases} J(x; \alpha, \beta) = \frac{2}{\sqrt{4\beta - \alpha^2}} \arctan \left( \frac{x\sqrt{4\beta - \alpha^2}}{2 + \alpha x} \right) \\ K(x; \alpha, \beta) = \ln(1 + \alpha x + \beta x^2) - \frac{\alpha}{2\beta} J(x; \alpha, \beta) \end{cases}$$

## RMI simulation configuration

- The simulations presented here were performed with the 2D Hytrac code.
- Planar geometry w/ AMR and front-tracking is used.
- Ideal gas EoS was used in all cases.
- Simulations were initialised using three regions:
  - Two materials in pressure equilibrium separated by a perturbed interface
  - A shocked region defined by the analytical solution to a 1D Riemann problem resulting in the required interface contact velocity  $u_c$
- Resolution (via AMR level), higher order numerical schemes and front-tracking methods were varied.
- The average perturbation amplitude was obtained by post-processing the bubble and spike features.
- Moving-window simulations, i.e., in the reference frame of the contact, were attempted but were found to yield unstable interfaces at low velocities ( $\sim 100$  m/s), as previously noted in [12].
- The simulation parameters are presented in Table 2.

Parameter	Tri-Lab test suite (Air-SF <sub>6</sub> )	Dimonte-Remington (Be-AGAR)	
		Slow	Fast
$\gamma_A$	1.276	1.80	
$\gamma_B$	1.093	1.45	
$\rho_A$	1.3 [kg/m <sup>3</sup> ]	1700 [kg/m <sup>3</sup> ]	
$\rho_B$	5.5 [kg/m <sup>3</sup> ]	120 [kg/m <sup>3</sup> ]	
$A^-$	0.616	-0.868	
$P$	909 [bar]	0.1 [MBar]	
$\epsilon$	0.441	0.991	0.998
$M_i$	1.3	9.1	18.2
$\lambda$	5.93 [cm]	100 $\mu$ m	
$h_0^-$	3.00 [mm]	10 $\mu$ m	14 $\mu$ m
$u_c$	97.0 [m/s]	35 [km/s]	71 [km/s]

Table 2: The parameters used to setup the Tri-Lab test suite and Dimonte-Remington simulations.

## Numerical results and analysis

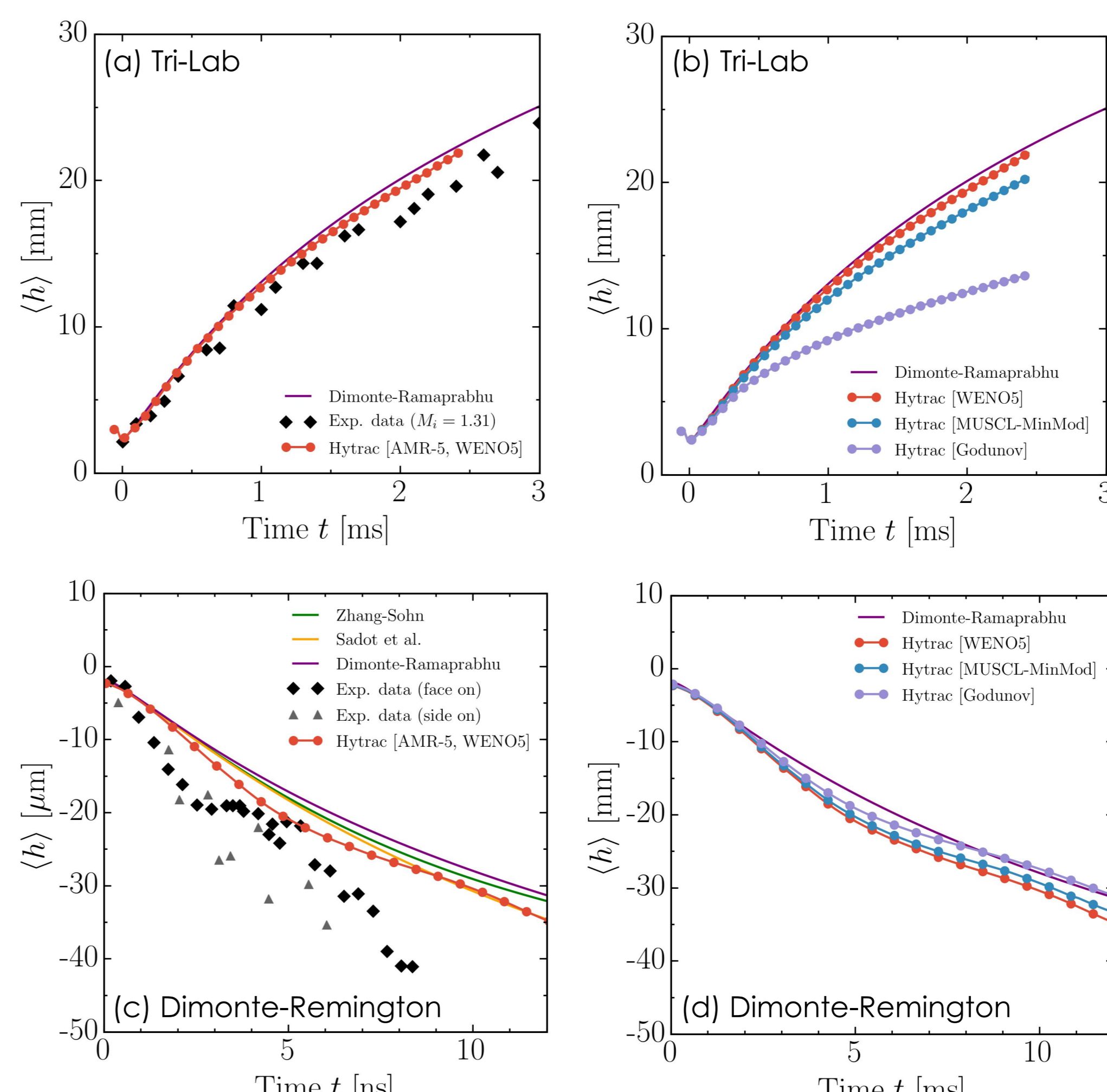


Fig. 5: (right) A comparison of the instability amplitudes for the TLTS and D-R cases as simulated using Hytrac, compared to the analytical models and exp. data. (a) The average amplitude from Hytrac using the WENO5 scheme compares favourably to the Dimonte-Ramaprabhu model and the shocktube data. (b) Higher order numerical schemes - Godunov (1st order), MUSCL (2nd order), WENO5 (5th order) - converge to the model more rapidly at fixed resolution. (c) For the D-R test, the Hytrac average amplitudes are in line with the models, but there is still some discrepancy from the exp. data. (d) Increasing order numerical schemes are compared for the D-R (fast) test. The variation is found to be smaller than for the TLTS case.

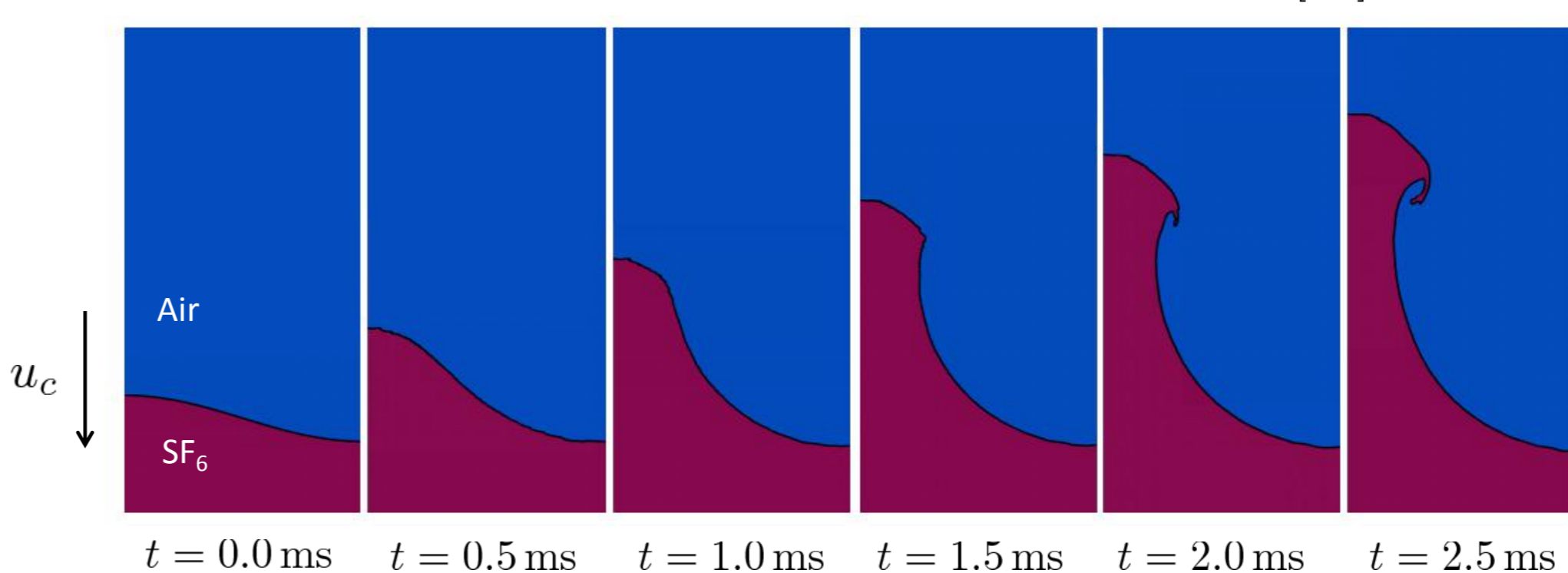


Fig. 6: (above) Time evolution of the instability for the Tri-Lab case. As shown in the above figure, these images correspond to Hytrac run with AMR-5, WENO5.

- The Hytrac RMI comparison for the TTS case is good at higher AMR levels using higher order numerical schemes. Both the Dimonte-Ramaprabhu model and experimental data are well matched.
- The effect of the front-tracking scheme on interface evolution is of continued interest.
- For the more extreme Dimonte-Remington conditions, Hytrac matches the analytical models initialised w/ the Richtmyer growth rate reasonably well, but there is a discrepancy with the experimental data.
- Matching the initial conditions for the radiative precursor is a challenging problem.

## Conclusions and future work

- We have performed Richtmyer-Meshkov instability simulations to validate Hytrac.
- The TLTS and D-R tests have been incorporated into the Hytrac CI suite.
- A good match to existing literature for both positive (Air-SF<sub>6</sub>) and negative (Be-AGAR) tests is found.
- This establishes trust in our modelling tools to evaluate novel, target-relevant parameters.
- Future steps: 1) validate the B code, 2) multi-mode RMI tests, 3) perform integrated target simulations, and 4) consider in-house experiments to provide new code validation data.

## References

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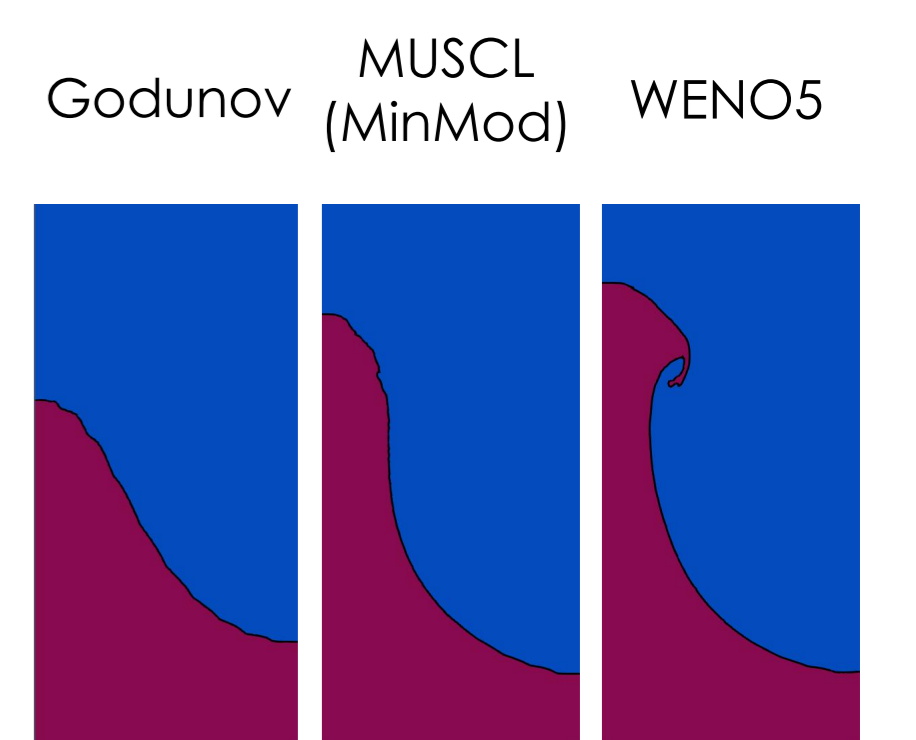


Fig. 7: (above) Interface comparison for the TLTS case after  $t = 2.5$  ms of simulation at  $n=10$ , AMR-5 ( $\Rightarrow dx=180 \mu$ m, i.e., 320 cells per  $\lambda$ ). The 'roll-up' feature appears for higher order numerical schemes.